

SUPPLEMENTAL TYPE CERTIFICATE

10016728 REV. 2

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

AIRBUS HELICOPTERS CANADA LIMITED

P.O.BOX 250
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FORT ERIE ON L2A 5M9
CANADA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.R.008

Type Certificate Holder: AIRBUS HELICOPTERS

Type: AS 350 / EC 130

Model: AS 350 B, AS 350 B1, AS 350 B2
AS 350 B3, AS 350 BA, AS 350 D

Original STC Number: TCCA SH97-15

Description of Design Change:

Installation of LH and/or RH Cargo Pods.

Rev. 1: Minor Changes embodied and new name of STCH.

Rev. 2: added Compatibility of a sliding door with the LH/RH Cargo Pod Installation for AS 350 B2 and AS 350 B3 helicopters only.

EASA Certification Basis:

The Type and OSD Certification Bases (CB) for the original product remains applicable to this certificate/ approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise database.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 10 August 2023

Fabrice LEGAY
Section Manager
Medium & Light Rotorcraft



Associated Technical Documentation:

Master Drawing List MDL-97-004 rev. I, TCCA approved.
Flight Manual Supplement FMS-ECL-9 Rev. 10, TCCA approved.
Instructions for Continued Airworthiness ICA-AHCA-294 Rev. 2, TCCA accepted

or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU/ Canada Bilateral Agreement.

Limitations/Conditions:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -